# STRUCTURAL ANALYSIS OF INFLATED MEMBRANES WITH APPLICATIONS TO LARGE SCIENTIFIC BALLOONS 

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## Overview

1. The balloon problem
2. Mathematical model for the analysis of partially inflated strained balloons
3. Analysis of pumpkin balloon

## The Balloon Problem: Design and Analysis

- Design - Determine the shape of a balloon to carry a payload of weight $L$ at a constant altitude.
- Typically, assume a statically determinate shape (consider balloon system weight and hydrostatic pressure).
- Actual balloon is constructed from long tapered flat sheets of thin film that are sealed edge-to-edge. Load tendons are attached along each seam.
- Analysis - Estimate film stresses.
- Model the balloon as an elastic membrane
- Include elastic reinforcing load tendons
- Consider launch, ascent, and float configurations.
- Mathematical model for the analysis of strained/partially inflated balloons supported by NASA Awards: NAG5-697, 5292, 5353.

Partially Inflated Balloons (same loading)


## Partially Inflated Balloon (Single Gore)



## Partially Inflated Balloon with Lobes



## Design Related Considerations

## Natural-Shape Equations ( $\sigma_{c}=0$ )

Axisymmetric membrane theory:
UMN, 1950s; further balloon development by J. Smalley, 1960-70s.


$$
\overrightarrow{0}=\frac{\partial}{\partial s}\left(r \sigma_{m} \mathbf{t}\right)-\sigma_{c} \mathbf{e}_{1}(\phi)+r \mathbf{f}
$$

$T(s)=2 \pi r(s) \sigma_{m}(s)$ - total meridional tension
$\mathbf{f}$ - $\underbrace{\text { hydrostatic pressure }}$ and film/tendon weight

$$
p=b z+p_{0}
$$

## Natural-Shape Balloons <br> Zero Pressure and Super-Pressure Designs



- Zero-pressure balloons ( $p_{0}=0$ ).

Typical missions are several days.
Open at base and need ballast to maintain constant altitude.

- Super-pressure balloon ( $p_{0} \gg b z_{\max }>0$ ).

Add sufficient pressure so that day/night volume changes are reduced.

## Super-Pressure Natural-Shape Balloon



A developable (ruled) surface
"Manufactured" design

- While the natural-shape design is axisymmetric, manufactured design consists of piecewise ruled surfaces.
- ZP-balloons can handle the film stresses that are normally encountered.
- With a natural-shape superpressure design, available thin films are not strong enough to contain the pressure, or too heavy, or too expensive.
- Solution: A pumpkin shape with very strong tendons.


## The Pumpkin Balloon



- Curvature in the hoop direction transfers load from film to the tendons.
- Increased tendon stiffness can be achived by tendon shortening
(there is a film/tendon mismatch!).


## Background on the Pumpkin Balloon

- J. Smalley coined the term pumpkin balloon. Extensibility of the film is used to achieve the pumpkin gore shape (early 1970s).
- CNES built several small pumpkin balloons, cutting half-gore panels with extra material (mid-late 1970s)
- Sewing techniques to gather material at gore seams
(N. Yajima, Japan, 1998, see Adv. in Space Res., 2000).
- NASA/ULDB - structural lack-of-fit (shorten tendons) + material properties (W. Schur, PSL/WFF, 1998, see AIAA-99-1526).
- There are several versions of the pumpkin balloon. We will analyze a NASA ULDB pumpkin design flown in 2001.


## Strain Analysis

## The Natural (unstrained) State of a Complete Balloon


$n_{g}=290$ for the ULDB we consider here.

## Observations and EM-Model Assumptions

- Linear stress-strain constitutive law
- Isotropic material ( $E$-Youngs modulus, $v$-Poisson's ratio)
- Constant strain model $\left(T \in S_{R e f} \longleftrightarrow \mathcal{T} \in \mathcal{S}\right)$
- Wrinkling via energy relaxation (Pipkin) - facets are taut, slack, wrinkled
- Energy relaxation allows a tension field solution
- Folds can be used to describe distribution of excess material.
- Load tendons behave like sticky linearly elastic strings
- Shapes are characterized by large deformations but small strains.
- Hydrostatic pressure is shape dependent


## Variational Principle for a Strained Balloon

|  | Problem ${ }^{\star}$ |  |
| :---: | :--- | :--- |
|  | For $\mathcal{S} \in \mathcal{C}$, |  |
|  | Minimize: | $E_{T}(\mathcal{S})=E_{P}+E_{f}+E_{t}+S_{t}+S_{f}$ |
|  | Subject to: | $V=V_{0}$ |
| $S$ |  |  |
| $\mathcal{S}$ | balloon shape |  |
| $E_{T}$ | set of allowable shapes |  |
| $V$ | Total energy |  |
| $E_{P}$ | Volume |  |
| $E_{f}$ | hydrostatic pressure potential |  |
| $E_{t}$ | gravitational potential energy due to film weight |  |
| $S_{t}$ | strain energy of tendons |  |
| $S_{f}$ | strain energy of film |  |

Problem * is discretized and solved by EMsolver - developed for balloon applications, written in Matlab (uses fmincon - find minimum of a nonlinear multivariable function with linear and/or nonlinear constraints).
Aspects of EM-model have been implemented in Ken Brakke's Surface Evolver.

## Energy Terms

Hydrostatic Pressure: $E_{P}=-\int_{\mathcal{V}} p d V=-\int_{S}\left(\frac{1}{2} b z^{2}+p_{0} z\right) \vec{k} \cdot d \vec{S}$,
Film Weight: $E_{f}=\int_{S} w_{f} z d A$
Tendon Weight: $E_{t}=\sum_{i=1}^{n_{s}} \int_{0}^{\ell_{d}} w_{t}^{i} z d s$
Tendon Strain: $S_{t}=\sum_{i=1}^{n_{s}} \int_{0}^{\ell_{d}} W_{c}^{*}\left(\dot{\gamma}_{i}\right) d s, \quad W_{c}\left(\dot{\gamma}_{i}\right)=\frac{1}{8} K_{t}\left(\left|\dot{\gamma}_{i}\right|^{2}-1\right)$.
Film Strain: $S_{f}=\int_{\Omega} W_{f}(\mathbf{G}) d A, \quad W_{f}(\mathbf{G})=\frac{1}{2} \mathbf{S}: \mathbf{G}$;
Strains: $\mathbf{G}=\frac{1}{2}(\mathbf{C}-\mathbf{I})$ - Green, $\mathbf{C}=\mathbf{F}^{T} \mathbf{F}$ - Cauchy; $\mathbf{F}$ - Def. Grad.
Second Piola-Kirchoff stress tensor

$$
\mathbf{S}(\mathbf{G})=\frac{t E}{1-v^{2}}\left(\mathbf{G}+v \operatorname{Cof}\left(\mathbf{G}^{T}\right)\right) .
$$

Fine wrinkling: replace $W_{f}$ by its relaxation $W_{f}^{*}$, allowing a Tension Field

## Energy Relaxation $\Longrightarrow$ Tension Field

In Pipkin's approach decompose M into three disjoint regions:
S - Slack region: Cauchy-Green strains are both negative, $\delta_{1}<0, \delta_{2}<0$;
T-Tense region: both principal stress resultants are positive, $\mu_{1}>0, \mu_{2}>0$;
U - Wrinkled region $(\mathrm{U}=\mathrm{M} \backslash \mathrm{S} \cup \mathrm{T})$.

$$
\begin{gathered}
\text { Classify each } T_{l} \in \Omega \\
W_{f}^{*}\left(\delta_{1}, \delta_{2} ; t, \nu, E\right)=\left\{\begin{array}{l}
0, \delta_{1}<0 \text { and } \delta_{2}<0 \\
\frac{1}{2} t E \delta_{2}^{2}, \mu_{1} \leq 0 \text { and } \delta_{2} \geq 0 \\
\frac{1}{2} t E \delta_{1}^{2}, \mu_{2} \leq 0 \text { and } \delta_{1} \geq 0 \\
\frac{t E}{2\left(1-v^{2}\right)}\left(\delta_{1}^{2}+\delta_{2}^{2}+2 v \delta_{1} \delta_{2}\right) \\
\mu_{1} \geq 0 \text { and } \mu_{2} \geq 0
\end{array}\right.
\end{gathered}
$$

*See FB and Collier, AIAA J, Vol 39, No. 9, Sept 2001, 1662-1672.



Principal Stresses: Superpresure Natural vs. Pumpkin


## Stress Analysis Summary

| $t=38 \mu \mathrm{~m}$ (1.5 mil) | Max Stress (stress resultant) |  |
| :---: | :---: | :---: |
| Tendon | Slack 2.9\% | Shorten 2.0\% |
| Meridional <br> Natural <br> Hoop | $78 \mathrm{MPa}(17 \mathrm{lbf} / \mathrm{in})$ $78 \mathrm{MPa}(17 \mathrm{lbf} / \mathrm{in})$ | 0 MPa ( $0 \mathrm{lbf} / \mathrm{in}$ ) <br> 5.25 MPa (1.41 lbf/in) |
| Meridional <br> Pumpkin <br> Hoop | $28 \mathrm{MPa}(6.09 \mathrm{lbf} / \mathrm{in})$ $40 \mathrm{MPa}(8.70 \mathrm{lbf} / \mathrm{in})$ | 0 MPa (0 lbf/in) <br> 4.25 MPa (0.92 lbf/in) |

## Conclusions

- Pumpkin design (shape + tendon shortening) offers a significant reduction in maximum stresses compared to natural-shape superpressure design.
- The variational formulation and optimization based solution process of EMsolver provides an analytical tool that is readily adaptable to other membrane and gossamer structures.


## Appendices

- (2002) Comparison of EMsolver predictions with measurements.
- Benchmark comparisons with ABAQUS
- (1998) Zero pressure natural shape;

EMsolver with virtual fold.

- (2001 - ) Spherical balloon with rope constraints;

EMsolver with strain energy relaxation.

## Compare EMsolver Predictions with Measurements

Joint work - Willi Schur (PSL/WFF); Tech. supp. - Roy Tolbert (NASA/WFF)

|  | Measured | Predicted | Absolute Error | Relative Error |
| :--- | :---: | :---: | :---: | :---: |
|  | $M$ | $P$ | $\|M-P\|$ | $\|M-P\| / M \mid$ |
| Diameter | 4.0606 | 4.034 | 0.0266 | 0.0064 |
| Z(Diam) | 1.2846 | 1.239 | 0.0456 | 0.0354 |
| Height | 2.4102 | 2.449 | 0.0388 | 0.0160 |

Set-up for test vehicle inflations: Elevation (el) and azimuth (az) were recorded.
(a) Side view - elevation measurements; a 4 ft ruler was attached to an overhead hoist and lowered until it was just touching the top of the balloon.
(b) Overhead view - azimuthal measurements, since it was difficult to locate the line of sight tangency point for az, the az-measurements are probably not as accurate as the el-measurements.
(a) Side view

(b) Top view


## Benchmarks: ABAQUS and EMsolver

1998 Zero-pressure natural shape balloon. Analyzed single gore. Joint work with W. Schur (PSL/WFF) for NASA Balloon Office

2001-present Spherical balloon with mooring ropes and rigid end caps. Joint work with Laura Cadonati (Princeton/MIT) for The Borexino Project (a solar neutrino particle detector experiment)

Comparison of
EMsolver (virtual fold, K. Brakke) and
ABAQUS (tension field, W. Schur)

ZP-natural shape Joint work with W. Schur (1998)

| Parameters |  |
| :---: | :---: |
| 159 gores | Gore length 182 m |
| $b=0.05429 \mathrm{~N} / \mathrm{m}^{3}$ | $V=0.82$ |
| $E=124 \mathrm{MPa}$ | $E_{t}=26.24 \mathrm{kN}$ |
| $m_{f}=18.7 \mathrm{~g} / \mathrm{m}^{2}$ | $m_{t}=0.0313 \mathrm{~g} / \mathrm{m}$ |
| $V=832515 \mathrm{~m}^{3}$ | (zero-slackness) |






## Borexino Containment Vessel (joint work with L. Cadonati Princeton/MIT)




Principal Stress Resultants, $P(z)=50 \mathrm{~Pa}$



## Principal Stress Resultants

Open System: $P(0)=96 \mathrm{~Pa}, P(2 R)=170 \mathrm{~Pa}$


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