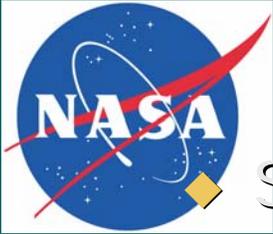


Design Analysis of A Scientific Balloon Instrument Support Structure (ISS)

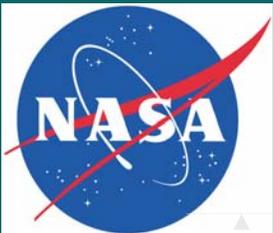
Victor Eyo
Mechanical Systems Branch (Code 548)
Wallops Flight Facility

Chip Frohlich
Swales Aerospace

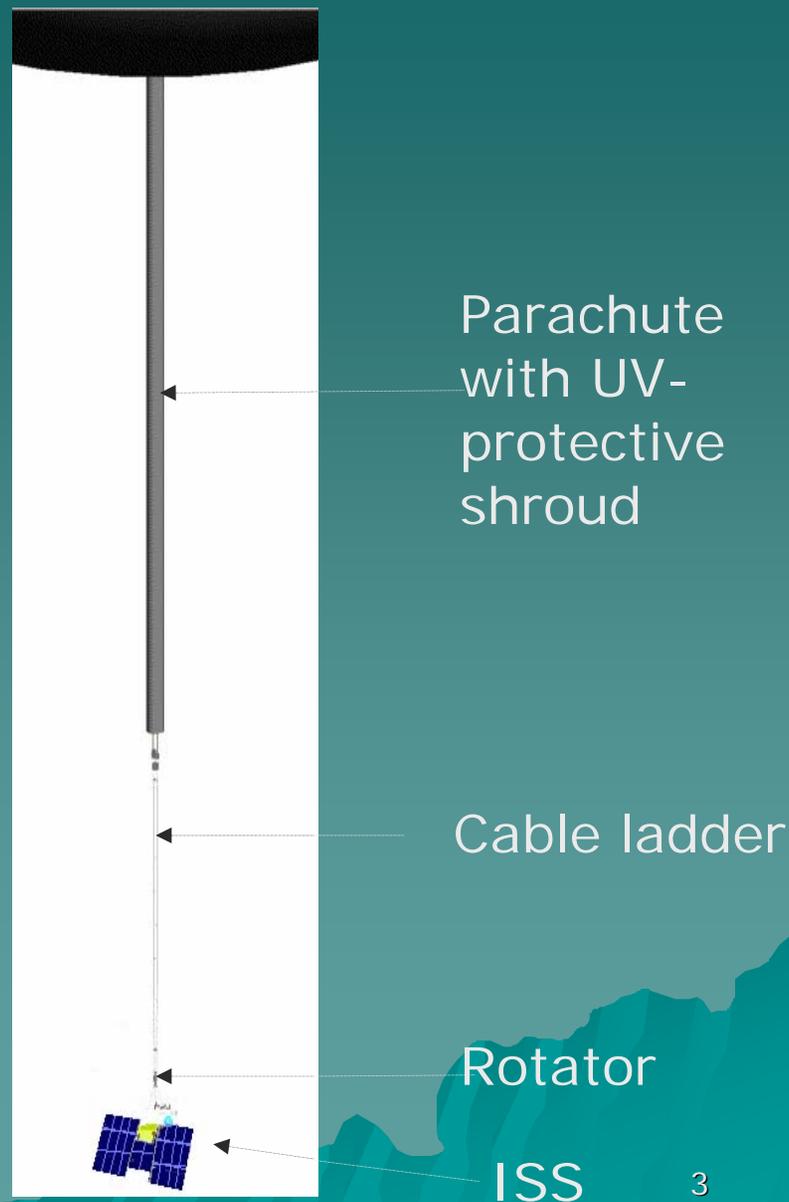
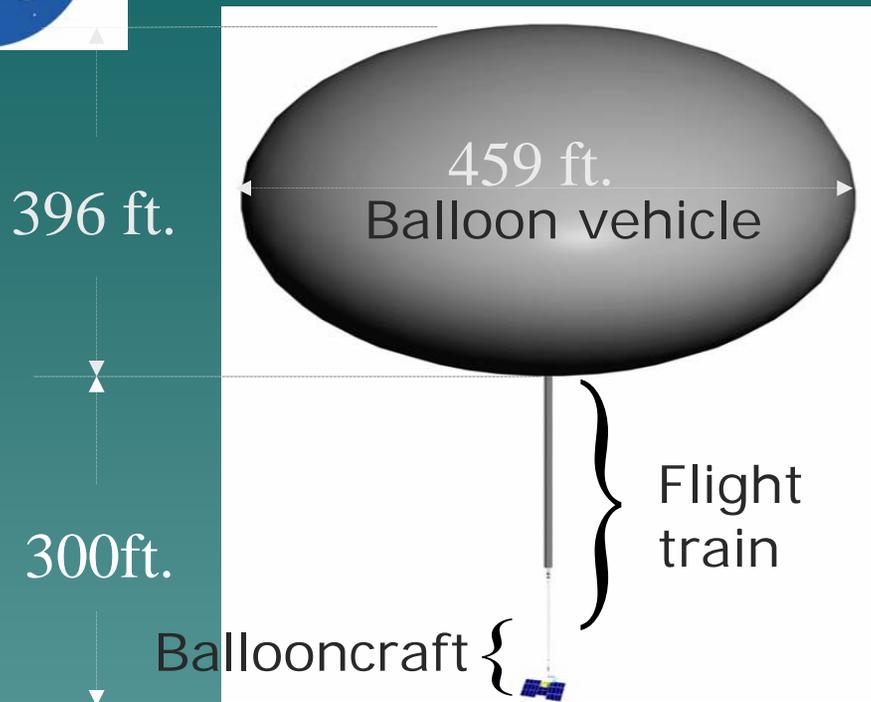


Motivation

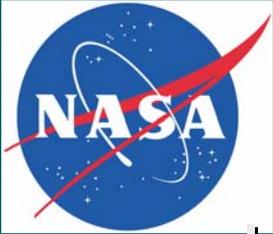
- Scientific Balloon ISS Analysis & Design
 - Simple structure
 - Becoming complex
 - ◆ Lightweight
 - ◆ Longer flight duration
 - ◆ Optimization – maintain integrity of science
 - Example using '04 CREAM campaign
- ◆ WFF Mechanical Systems Branch (Code 548)
 - Broad support services in the areas of:
 - ◆ Structural analysis
 - ◆ Mechanical design
 - ◆ Thermal engineering
 - ◆ Materials research and development
 - ◆ Assembly, integration and testing
 - Balloons Program Office
(<http://www.wff.nasa.gov/~code820/>)
 - <http://www.wff.nasa.gov/about/capabilities.php>



Scientific Balloons



- ◆ Volume: 39.57mcf
- ◆ Weight: 4,055lb
- ◆ Float Altitude: 125Kft
- ◆ Payload: 4,153lb
- ◆ Film: 20.32 μ m (0.8mil)



The Instrument Support Structure

◆ ISS

- Assembled set of mechanical components that provide the primary load path to support the weight and inertial loads of the

- ◆ science payload

- ◆ ballooncraft support systems

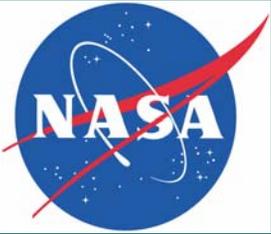
◆ The National Scientific Balloon Facility (NSBF), Palestine, TX

- Specifies loading conditions

- Responsible for flight worthiness certification of the ballooncraft

- Responsible for flight operations





CREAM ISS Initial Design

◆ Cosmic Ray Energetics And Mass Experiment

– Dec 2000

◆ In

◆ Co

– Initial

– MSC

◆ 42

◆ To

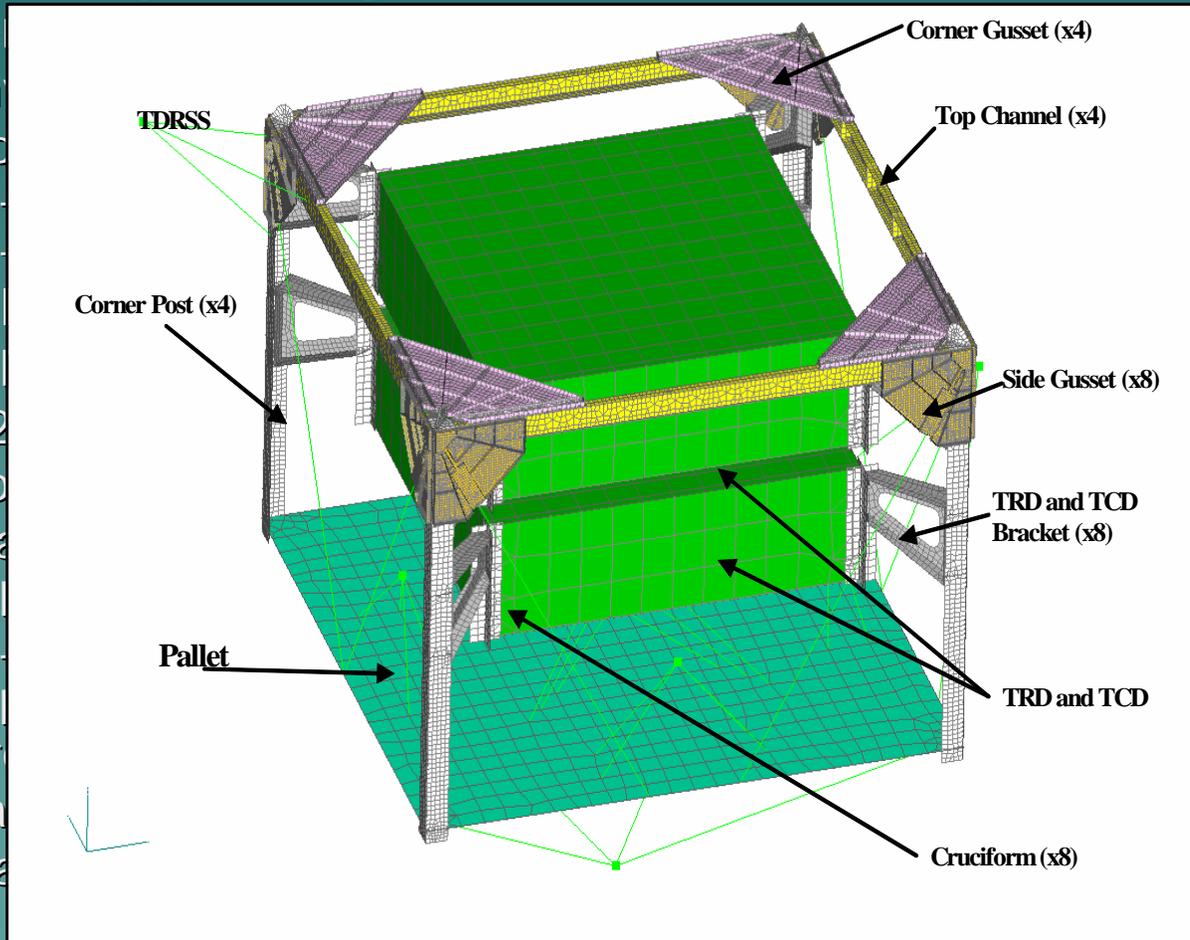
◆ Pla

◆ All

◆ All

◆ Fa

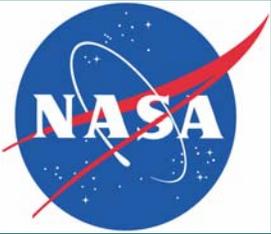
◆ Ma



5×10^{14} eV)

ty
and Mexico

al
modified plate

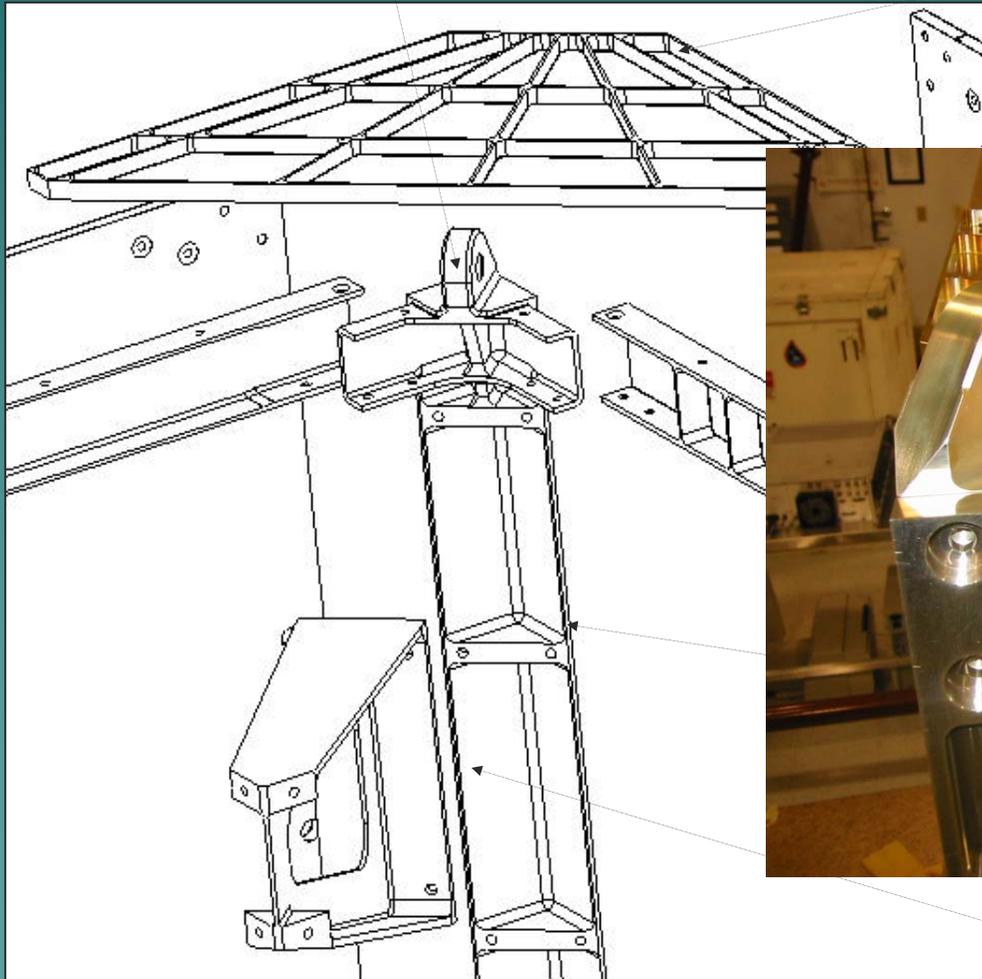


CREAM ISS Initial Design

Cable clevis interface

7075 corner gusset

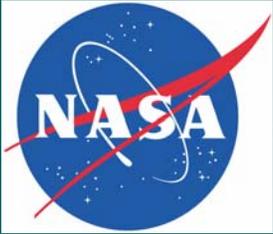
Radiator panels bolt to top channels and corner post w/ clip



channel
and
solar

el
os

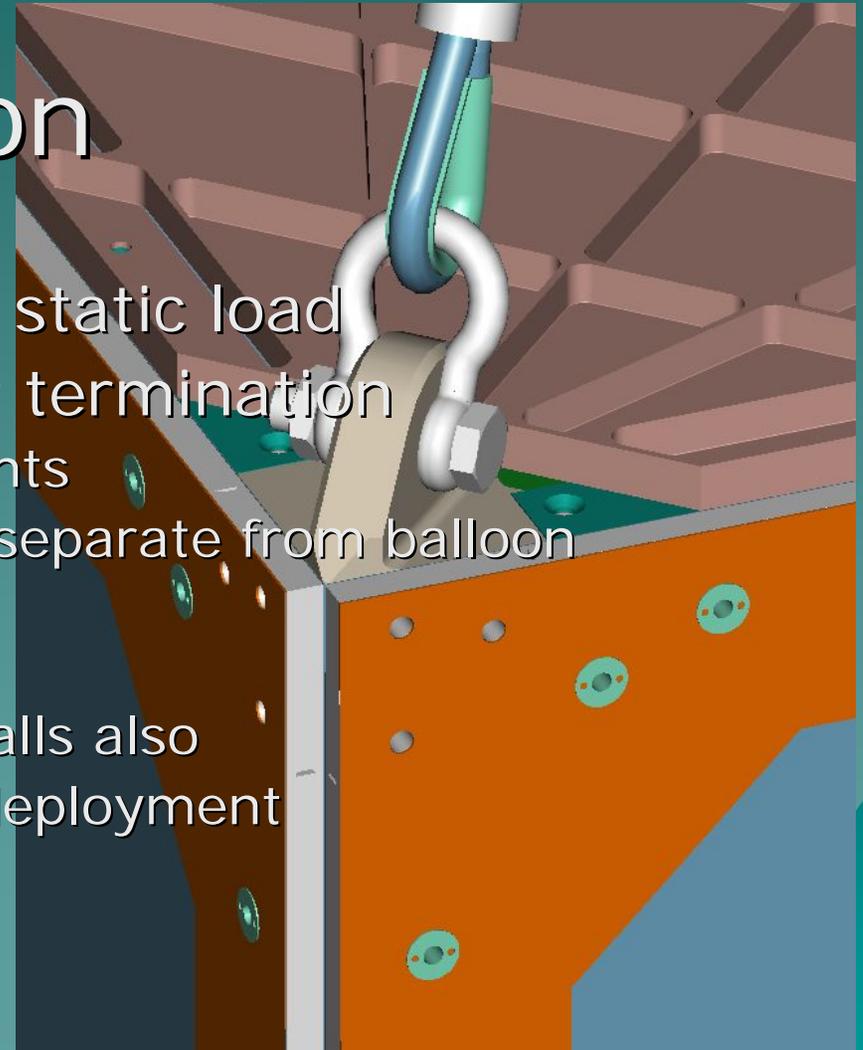
(8) TRD brackets weight optimized

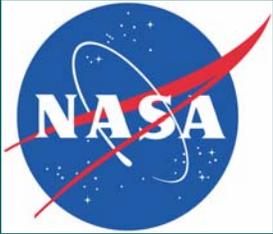


CREAM ISS Initial Design

◆ Loading Condition

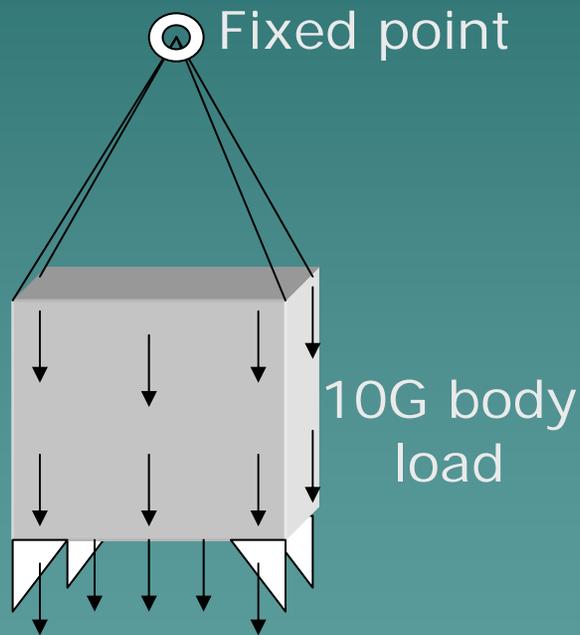
- Transient
- Applied analytically as a static load
- Imparted to structure at termination
 - ◆ End of scientific experiments
 - ◆ Ballooncraft & flight train separate from balloon vehicle
 - ◆ Balloon free-falls to earth
 - ◆ Rest of ballooncraft free-falls also
 - ◆ Shock load at parachute deployment



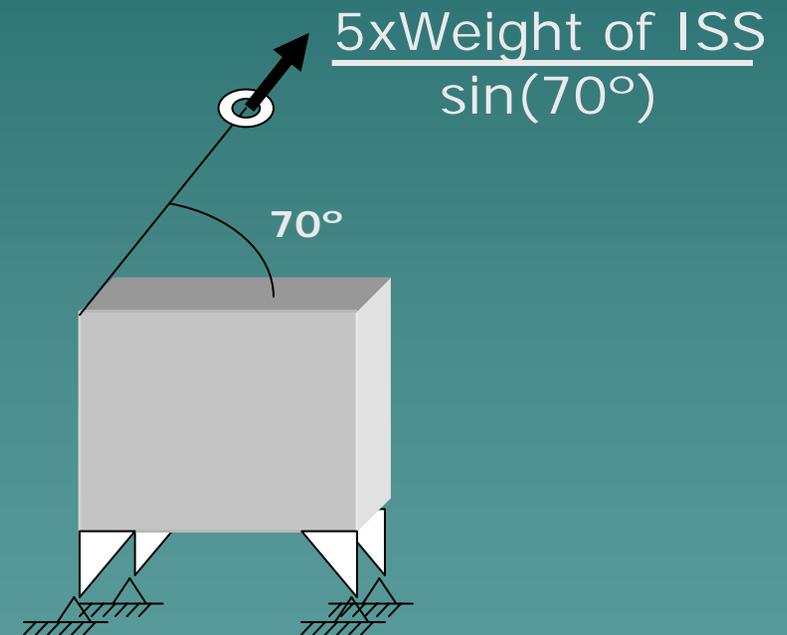


ISS – Loading Condition

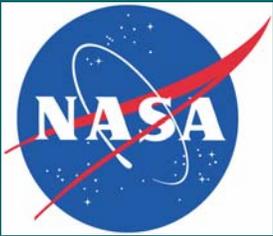
- ◆ The ISS design is such that all load-carrying structural members are capable of withstanding the NSBF prescribed conditions without **ultimate structural failure**.



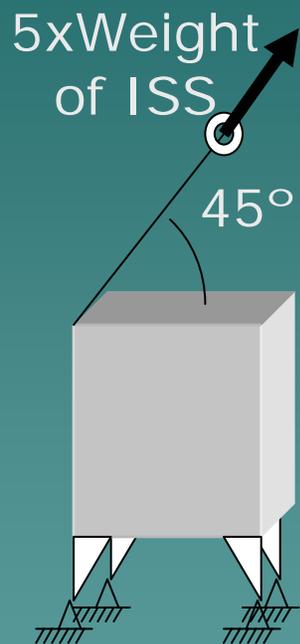
Assumes the ISS falls straight down and is caught by the parachute.



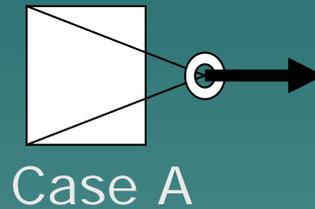
5x safety factor standard for overhead lifting with cables. The load path through the ISS should be as strong as the cable.



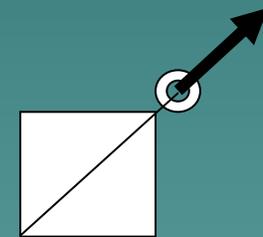
Initial Design – Loading Condition



Top View

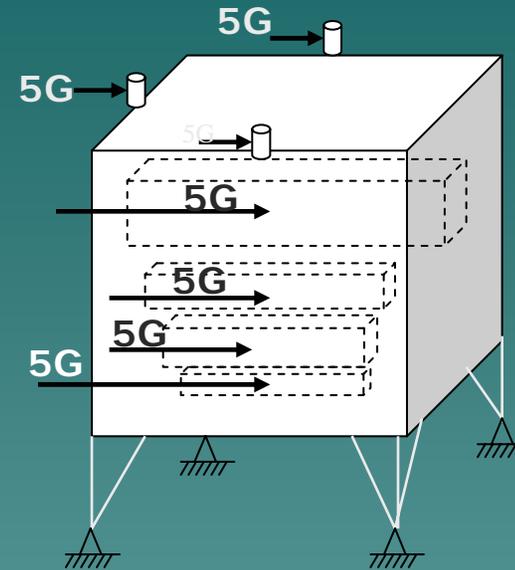


Case A

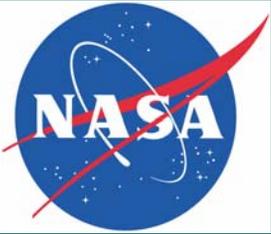


Case B

Assumes that the ISS rotates 45° as it falls and is caught by only one or two cables.

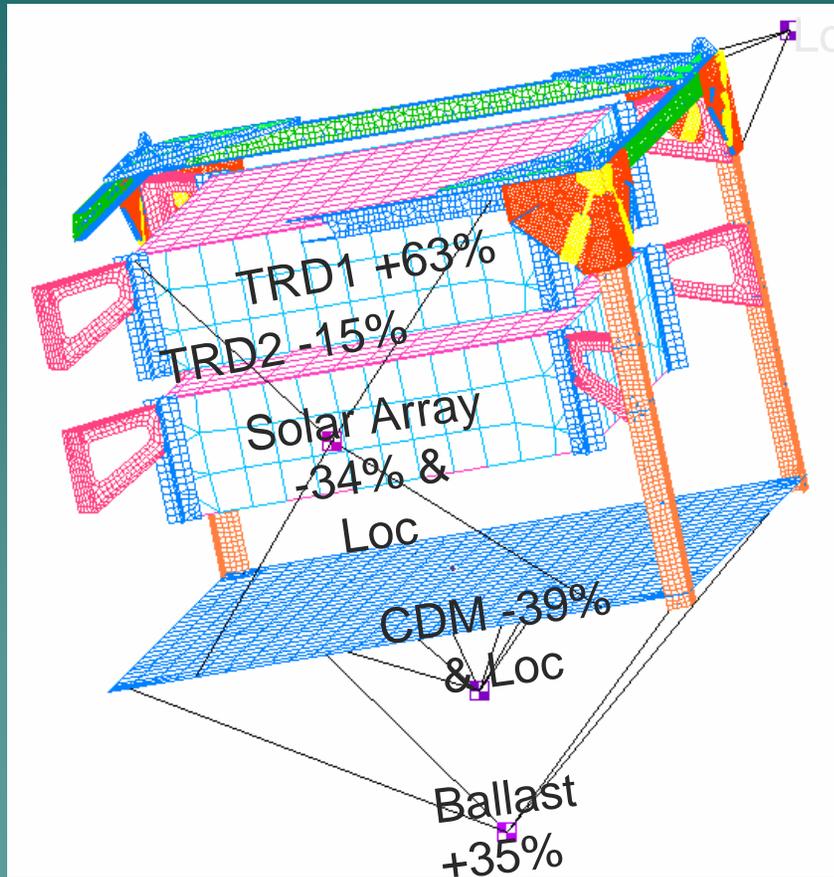


Assumes that all components attached to the ISS will remain attached in the event that the ISS rotates as it is caught by the parachute and inertial side loads are applied to components attached to the ISS



CREAM ISS Evolution

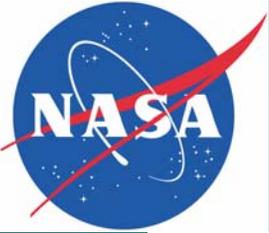
TDRSS +82% &



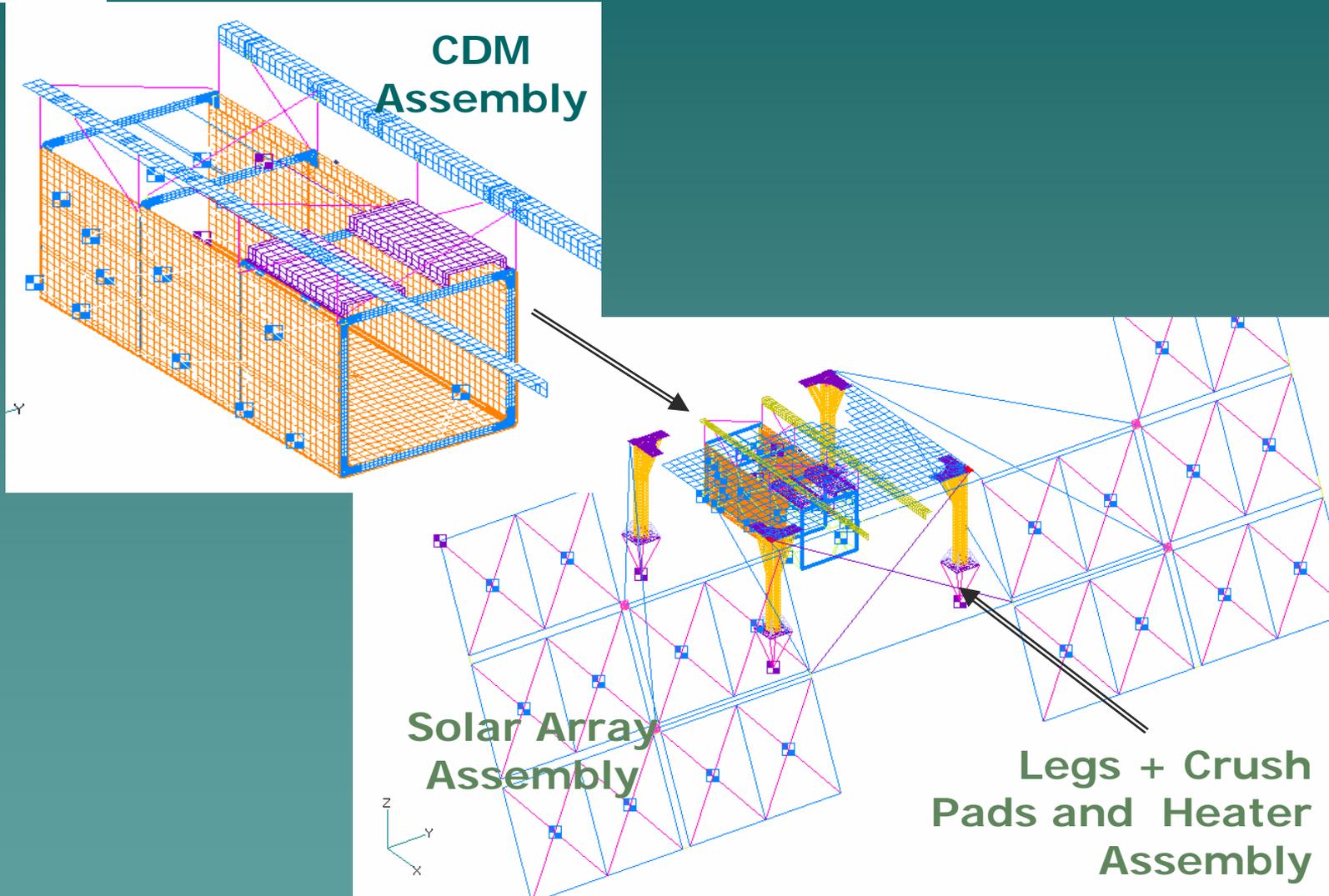
erred to WFF

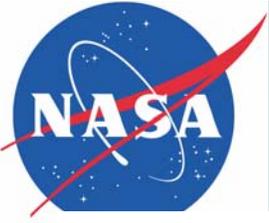
delivered to WFF

EnglishUnits (in/lb/s)	2003 ISS	2004 ISS
Mass	13.524	13.387
X_{cg}	2.592	-1.387
Y_{cg}	0.208	0.375
Z_{cg}	0.636	-8.909
I_{XX}	8,292.05	23,041.16
I_{YY}	10,120.71	22,310.28
I_{ZZ}	6,067.43	10,190.67
I_{XY}	-138.63	-241.33
I_{YZ}	133.84	263.22
I_{ZX}	651.04	-137.59

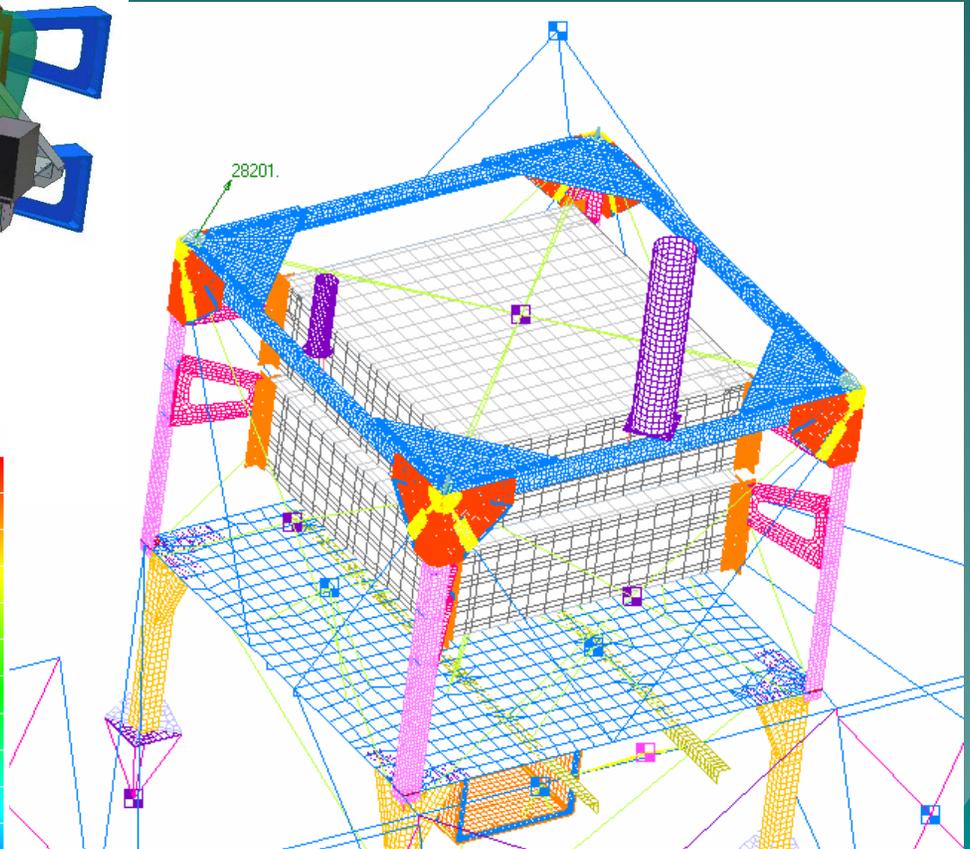
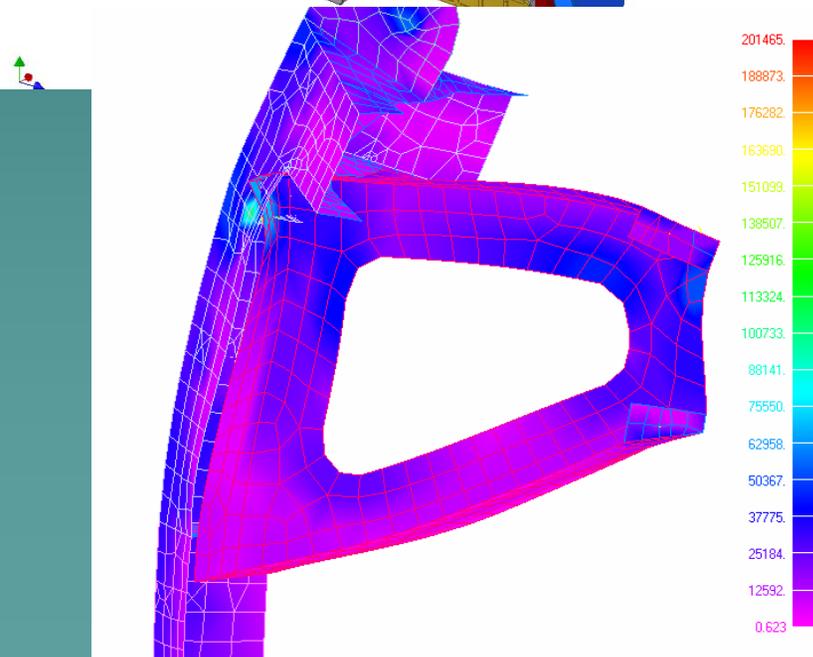
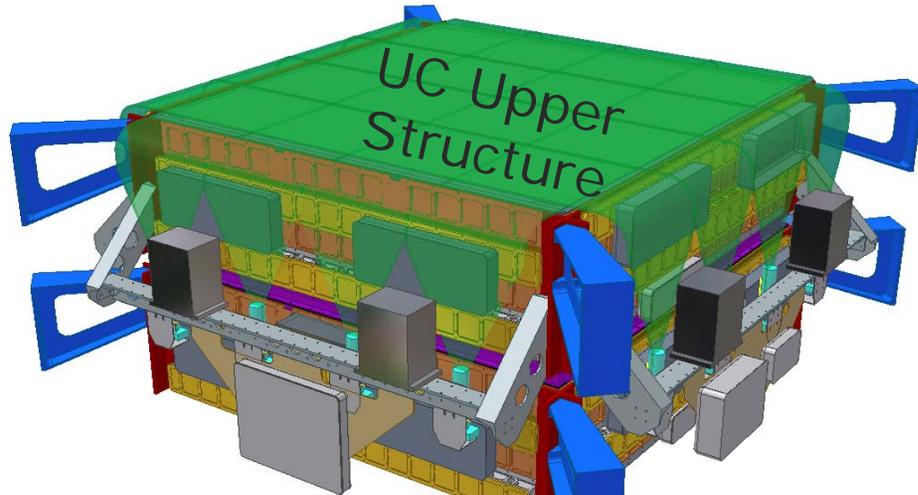


CREAM ISS Design Update

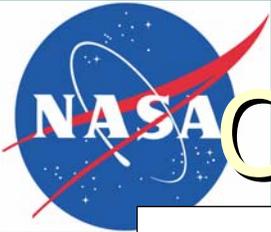




CREAM ISS Design Update



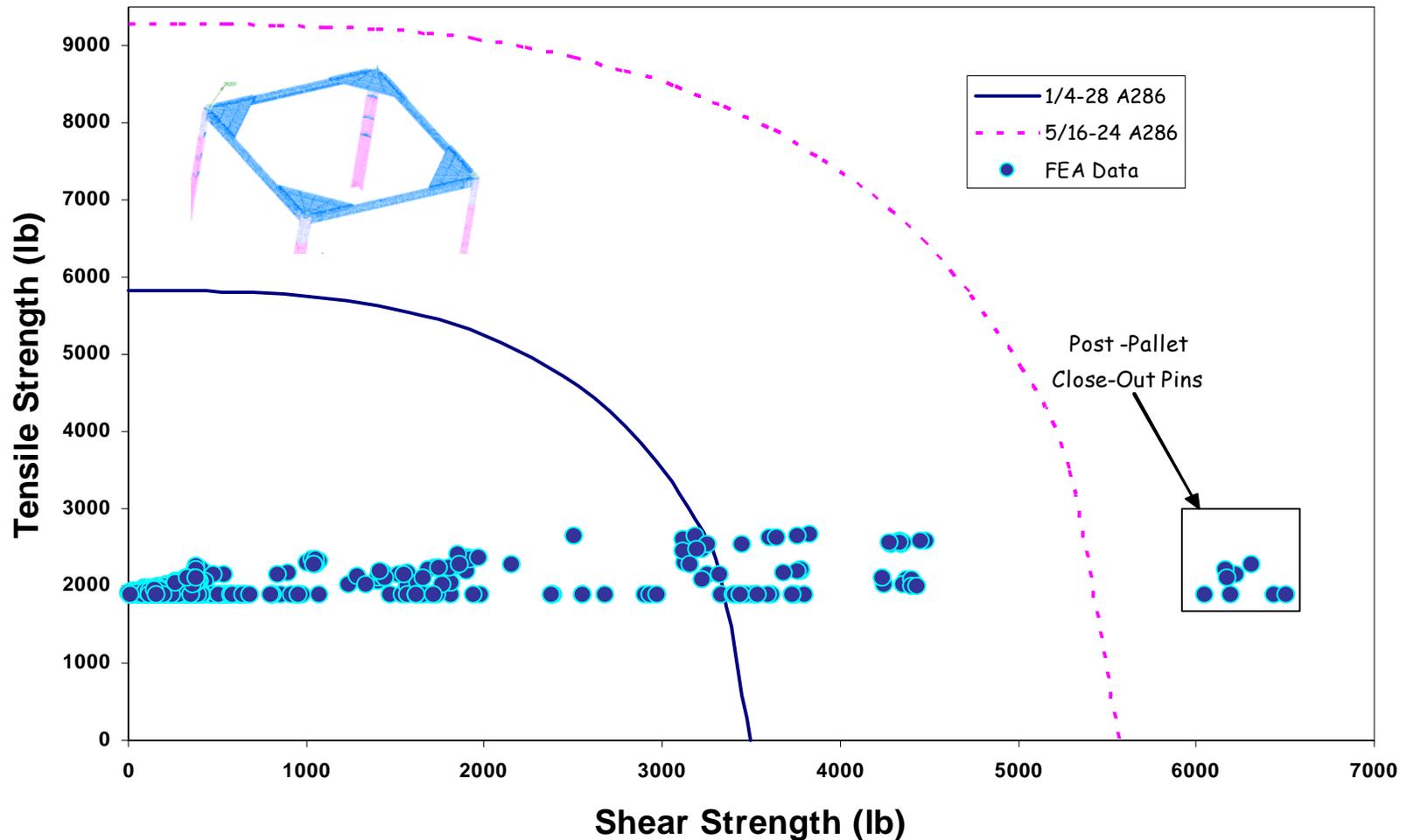
- ◆ Problem #1 - Big
– Massive fastener failure

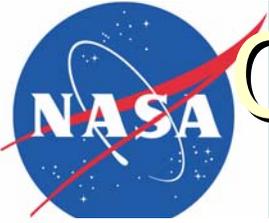


CREAM ISS Design Update

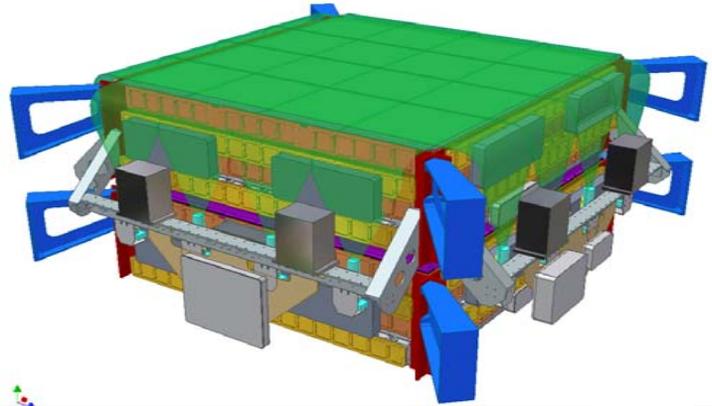
Interaction Curve

1/4-28 A-286 Bolt





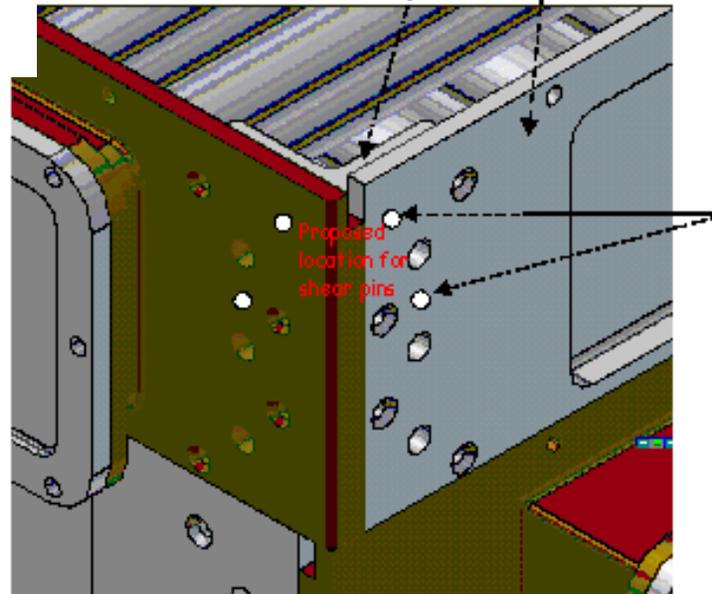
CREAM ISS Design Update



- ◆ Fasteners inaccessible for replacement

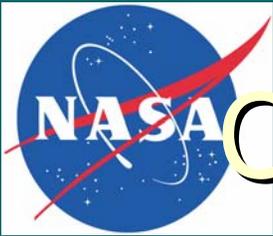
Pins will not penetrate this angle bracket,

but will go through the cruciform and the outer walls of the upper TRD.



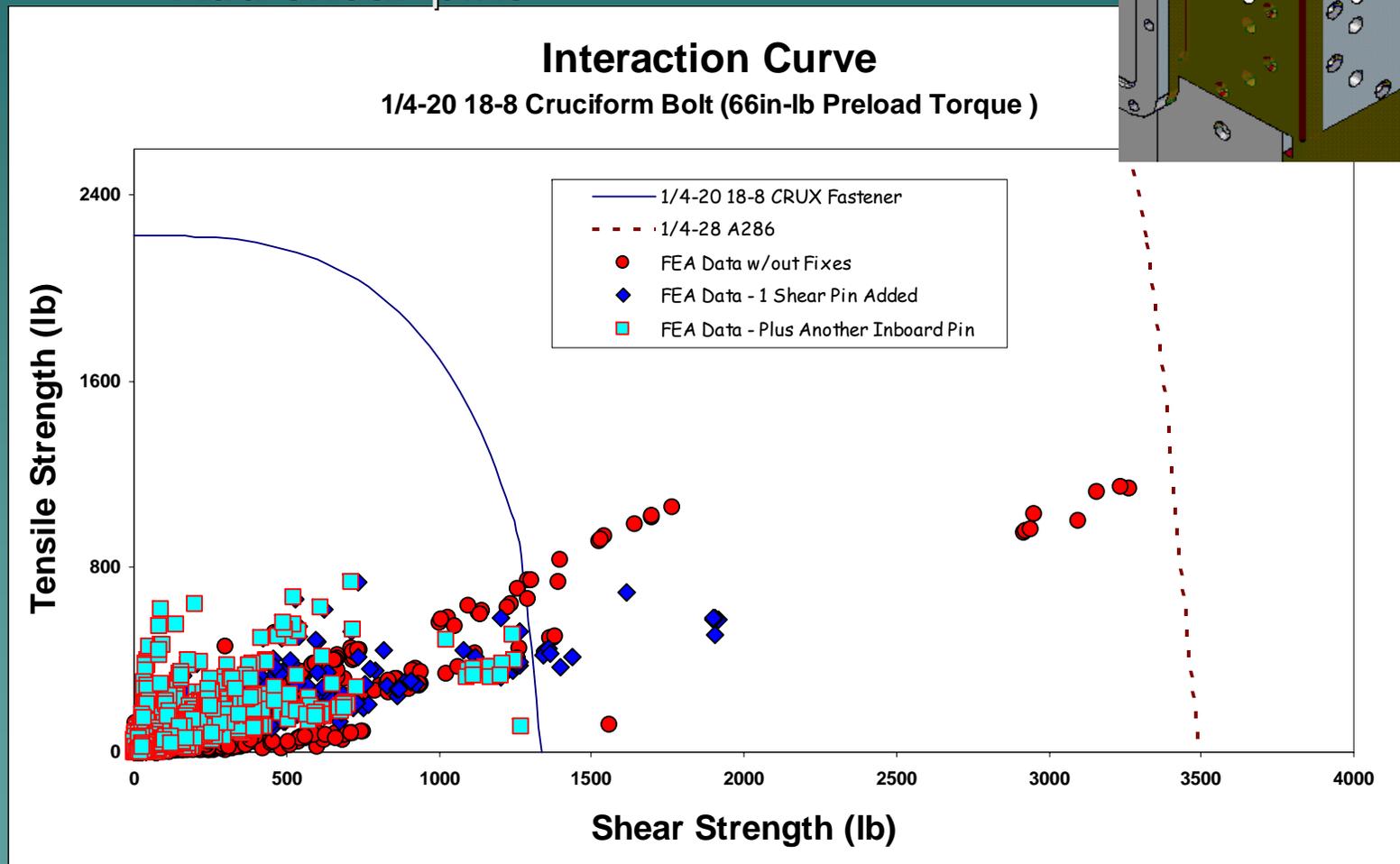
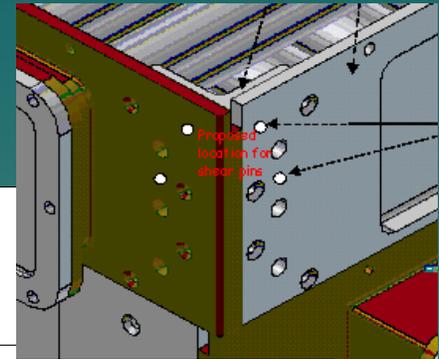
Proposed location for shear pins (x4 per corner, 1-in line vertically, the other above topmost fastener, inline with flat head screw.)

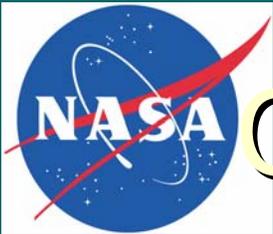
Section of Corner of Upper TRD Assembly



CREAM ISS Design Update

- ◆ Solution
 - Add shear pins



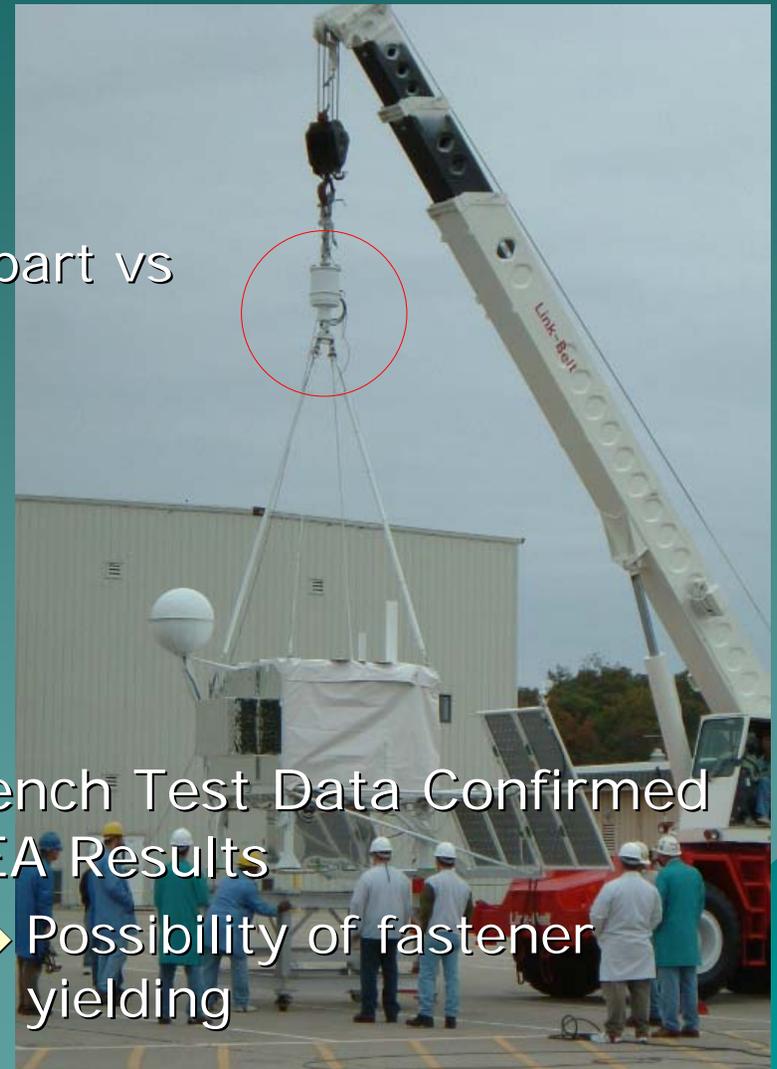


CREAM ISS Design Update

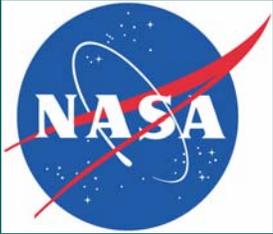
- ◆ Problem #2 - Bigger
 - Single point failure
 - Flight tested
 - Conflict: Flight experience with part vs FEA data



Lower Tri-Plate

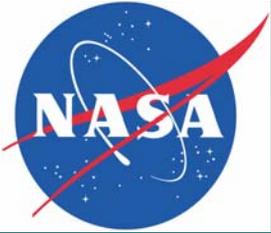


- Bench Test Data Confirmed FEA Results
- ◆ Possibility of fastener yielding

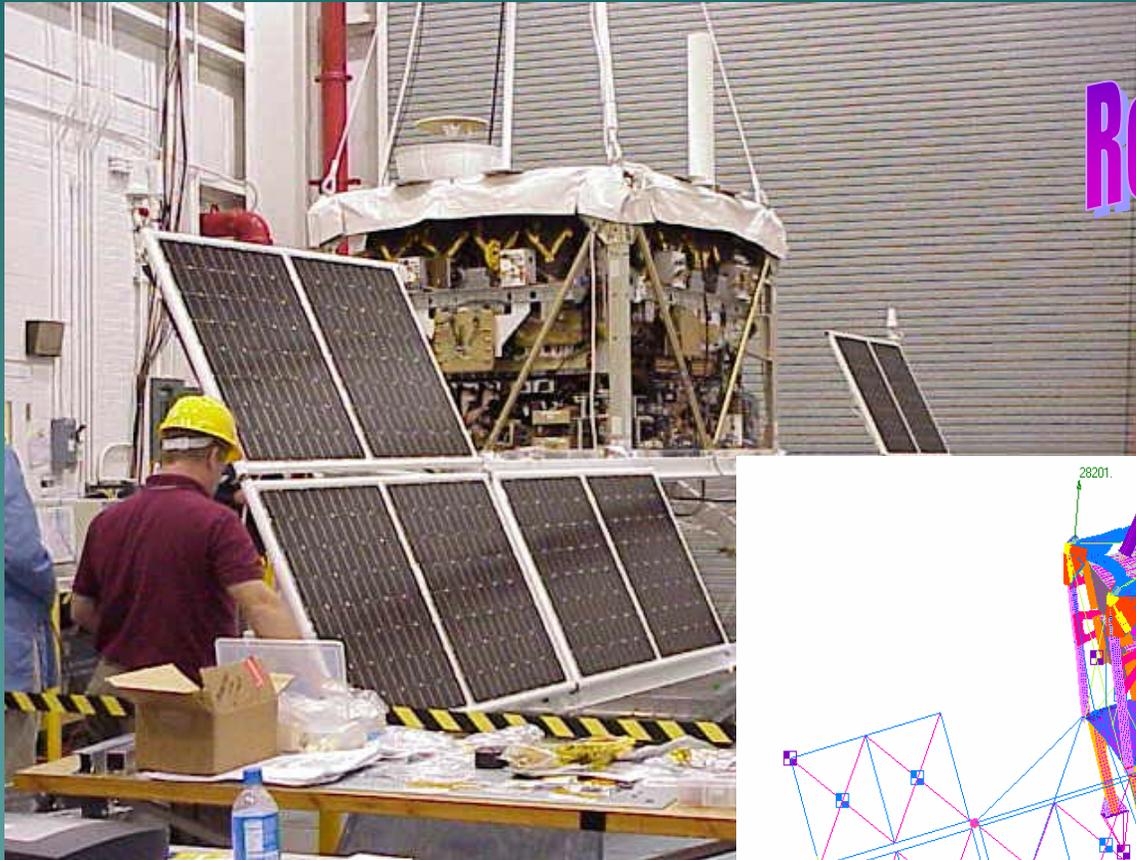


ISS Design & Analysis Challenges

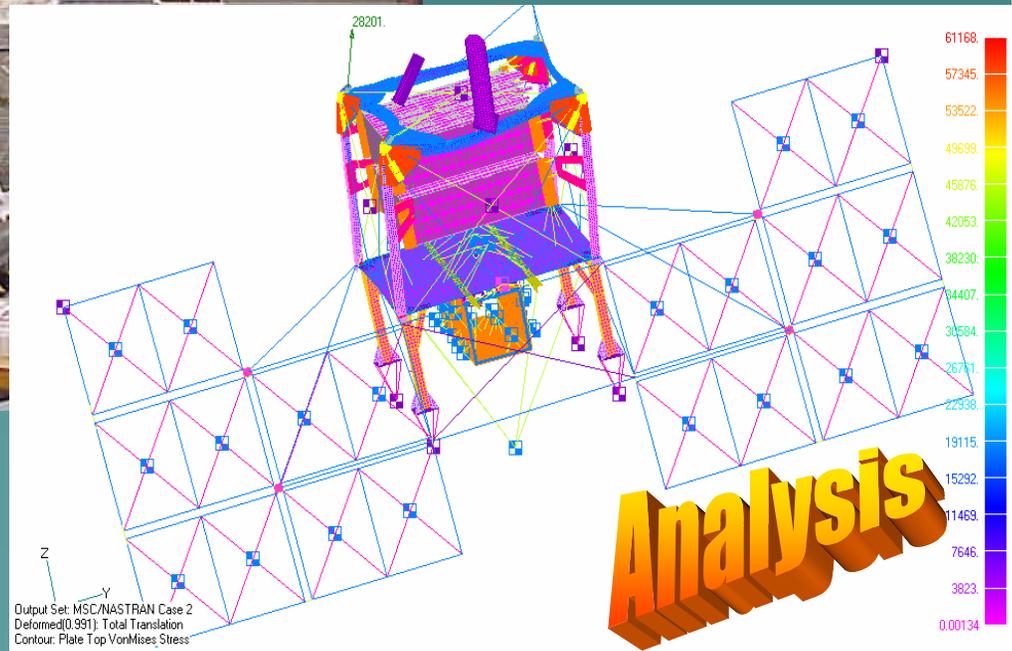
- ◆ What failure criteria to apply?
 - Yield/ultimate
- ◆ Quick turn-around flight
 - Re-flight → aborted primary flight
 - Re-flight → opportunity
 - Post yield & post buckling analyses
- ◆ System Constraints
 - ULDB vs LDB missions
 - ISS boundary conditions
- ◆ Loading Condition
 - Better definition of termination loads



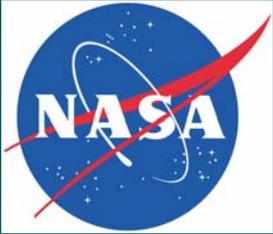
CREAM ISS - Images



Real Life



Analysis

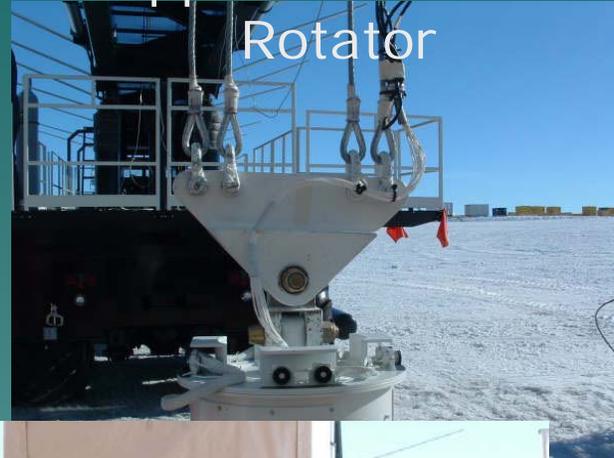


CREAM ISS - Images

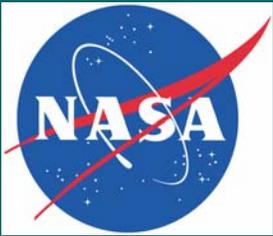
Lifting Exercises



Upper Tri-Plate & Rotator



Men @ Work



CREAM ISS - Images

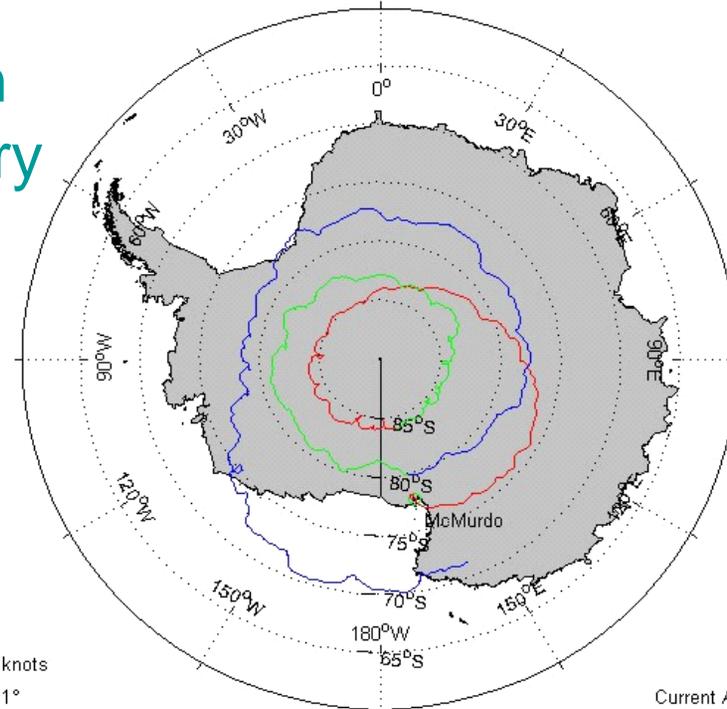
Balloon Inflation



Lift-Off

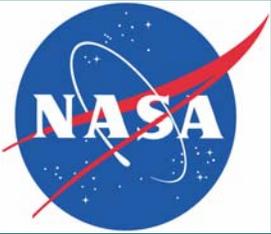
Balloon Trajectory

CREAM Flight Data: Trajectory
Covering period from: 2004-12-15 23:22:56 to 2005-01-27 02:00:31



Current Speed: 17.2 knots
Current Course: 128.1°
Current Lat: -71°17'3.72"
Current Lon: 157°52'54"

Current Altitude: 13828.7402 feet
Current MET: 41 days 21 hrs 31 mins 30.783 sec since launch
Current Time: 2005-01-27 02:00:31 UTC



CREAM ISS - Images

Landing & Recovery



Good Impact Landing



Ouch!!! Just Hacking Away at The ISS